

**FORMAC PROGRAM TO ASSIST IN THE ANALYSIS OF LINEAR CONTROL
SYSTEMS USING STATE VARIABLE FEEDBACK DESIGN TECHNIQUE**

By Charles R. Slivinsky, Lois T. Dellner, and Dale J. Arpasi

**Lewis Research Center
Cleveland, Ohio**

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

For sale by the Clearinghouse for Federal Scientific and Technical Information
Springfield, Virginia 22151 - CFSTI price \$3.00

FORMAC PROGRAM TO ASSIST IN THE ANALYSIS OF LINEAR CONTROL SYSTEMS USING STATE VARIABLE FEEDBACK DESIGN TECHNIQUE

by Charles R. Slivinsky*, Lois T. Dellner, and Dale J. Arpasi

Lewis Research Center

SUMMARY

A recently developed technique for control system design utilizes the feedback of all the state variables of the system. A program written in FORMAC to facilitate this design procedure is discussed.

The program must be supplied the open-loop transfer function, properly segmented into as many parts as the order of the system. Algebraic expressions for the unknown state variable feedback coefficients (the gains through which the state variables are fed back) are then calculated. Under certain conditions that result in a set of n algebraic equations linear in the unknown feedback gains, the program accepts additional input (the transfer function describing the desired system response) and calculates the values of the feedback gains which yield this response. A root locus diagram is plotted to display system sensitivity to variations of the gain in the left-most block of the open-loop transfer function.

Several examples of the use of the program are included and a complete program listing is given in an appendix.

INTRODUCTION

An important conclusion of modern control theory is that to minimize a broad class of integral type performance indices all the state variables of the system to be controlled must be fed back (refs. 1 to 3). Recently, this conclusion has been used as the basis for a new method of linear system synthesis, called state variable feedback design technique (ref. 1). Although the method has its origin in modern control theory, the tools used are familiar to engineers acquainted only with conventional control theory - namely, the Laplace transform and the frequency domain.

* Graduate Associate in Research, University of Arizona, Tucson, Arizona.

A system state variable is one of a set of n linearly independent variables, which are sufficient to completely describe the behavior of the system. These variables may be time derivatives of an output variable of the system (phase variables) or actual physical variables which may or may not be measurable.

The object of the design procedure is to achieve an initially specified closed-loop response by the proper selection of the gains of amplifiers through which the state variables of the system are fed back. The task is accomplished by formulating and solving a set of algebraic equations for the numerical values of the feedback amplifier gains. For control systems whose order exceeds two or three, a good deal of effort must be expended in carrying out the computational aspects of the design. However, the nature of the synthesis method is such that a digital computer can be used to perform much of the tedious calculation. A computer program for performing this function has been developed and utilized for state variable feedback design of several servosystems.

During the following discussion, some background concerning state variable feedback design technique is given. However, it is assumed the reader will refer to references such as 1 and 4 to obtain a full understanding of the method. Our objective is not to give a thorough treatment of the design technique but to provide information pertinent to the use and limitations of the computer program.

SIMPLE EXAMPLE OF STATE VARIABLE FEEDBACK DESIGN TECHNIQUE

To initiate the discussion of both the state variable feedback design technique and the computer program, consider the system whose block diagram is shown in figure 1(a). Here, quantities designated by capital letters represent Laplace transformed quantities, that is, functions of the complex frequency variable s . The system input is $E(s)$, the state variables are $X_1(s)$, $X_2(s)$, and $X_3(s)$, and the system output is $C(s) = X_3(s)$. In the left-most block is the closed-loop gain k . For the sake of clarity, no physical interpretation is given to the variables. Note that the outputs of each of the blocks can be taken as the state variables; in other words, the natural system variables are a suitable choice for state variables.

The open-loop transfer function of the system in figure 1(a) is

$$G(s) = \frac{k(s + 3)}{s(s + 2)(s + 4)}$$

Through the use of the state variable feedback design technique, the closed-loop transfer function ($C(s)/R(s)$) of this system may be made to correspond to any comparable desired transfer function $G_{des}(s)$. The design technique requires that all state variables

be fed back through linear amplifiers as shown in figure 1(b); the gain of each amplifier (feedback coefficient) is then determined so that $C(s)/R(s) = G_{des}(s)$.

Suppose, in this case, the desired closed-loop response is

$$G_{des}(s) = \frac{14.1667(s + 3)}{(s + 5)[(s + 1.5)^2 + (2.5)^2]} \quad (1)$$

$$= \frac{14.1667(s + 3)}{s^3 + 8s^2 + 23.5s + 42.5}$$

To solve for the feedback coefficients (h_1, h_2, h_3) that allow the desired closed-loop transfer function $G_{des}(s)$ to be achieved, it is first necessary to calculate the expression for $C(s)/R(s)$ for figure 1(b). This task is facilitated by block diagram manipulation to give the block diagram shown in figure 1(c). In this figure the forward path transfer function $G(s)$ is unaltered. The feedback transfer function $H_{eq}(s)$ is the single feedback path transfer function which is equivalent to the multifeedback path arrangement of figure 1(b).

With

$$\frac{C(s)}{R(s)} = \frac{G(s)}{1 + G(s)H_{eq}(s)} \quad (2)$$

the closed-loop transfer function is found to be

$$\frac{C(s)}{R(s)} = \frac{k(s + 3)}{s^3 + [kh_1 + kh_2 + 6]s^2 + [4kh_1 + 3kh_2 + kh_3 + 8]s + 3kh_3} \quad (3)$$

In order for equations (1) and (3) to be identical, the gain k must be 14.1667 and the following equations must be satisfied:

$$kh_1 + kh_2 + 6 = 8$$

$$4kh_1 + 3kh_2 + kh_3 + 8 = 23.5$$

$$3kh_3 = 42.5$$

The feedback coefficients are, therefore,

$$h_1 = -0.329$$

$$h_2 = 0.471$$

$$h_3 = 1.000$$

To illustrate system sensitivity, a root locus diagram may be plotted.

A review of the preceding example will indicate the need for mechanizing the algebraic manipulation and numerical calculations necessary to use the state variable feedback design technique in large problems. Manual calculation of the feedback gains is tedious and open to error, even for as simple a system as discussed here. Moreover, specification of a desired closed-loop transfer function is not usually a simple task. Aspects other than the overall response of the system must be weighted in the control design. For instance, it may be desirable (1) to keep the feedback gain of a certain state variable to a minimum in the hope of eventually eliminating the necessity of measuring or generating it, or (2) to reduce system sensitivity to gain changes at some unknown distribution of the desired closed-loop poles. These aspects would require the specification of the $G_{des}(s)$ on a trial and error basis and the calculation of the feedback gains for each trial. The volume of calculation required for such an approach makes the adaptation of the design procedure to a computer almost mandatory.

PROBLEM PREPARATION

Application of the state variable technique to control system design requires that (1) the open-loop transfer function $G(s)$, describing the system to be controlled, be defined, and (2) the transfer function $G_{des}(s)$ of the desired response be specified. The use of the FORMAC program as an aid in this application places certain requirements on the form of this input data. The transfer function $G(s)$ must be expressed as

$$G(s) = \prod_{i=1}^{i=n} K_i G_i(s)$$

This means the block diagram defining the problem must have been drawn as in figure 2 where

- (1) The state variables are $X_i(s)$.
- (2) The constants are K_i .

(3) The rational functions $G_i(s)$ with constant coefficients are defined so that (a) the coefficient of the highest degree term in each denominator is unity and (b) the h_i are constant but unspecified gain coefficients.

The reader may object to the exclusion of polynomials in s from the feedback blocks because it is often impossible or not economically feasible to measure all the state variables of a system, but it is usually possible to generate them. Such a situation might arise for the system of figure 1(b). Here, for instance, it may not be possible to measure $X_2(s)$, but the variable can be generated from $X_1(s)$ or $X_3(s)$ as in figure 3(a) or (b). Similarly, it may be necessary to generate the state variable $X_1(s)$, as shown in figure 3(c). All of these cases can be made to conform (to the restriction placed on the h_i 's) by block diagram manipulation. The case where one or more of the G_i contains a pair of complex poles can also be handled; the design of the fuel valve control system for a jet engine given in appendix A is such a case.

If the program is to continue beyond the formulation (and output) of the numerator and denominator polynomials of $H_{eq}(s)$ and $C(s)/R(s)$ to the solution for the numerical values of the h_i 's, it is required that the contents of the first block $G_1(s)$ be of the form $k/(s + c)$ where c is a specified constant and k is not specified.

There are also restrictions on the form of $G_{des}(s)$:

(1) The numerator of $G_{des}(s)$ must be exactly the same as the numerator of $G(s)$ except that all constants must be specified in $G_{des}(s)$. One unspecified constant k must be contained in $G_1(s)$.

(2) The degree of the denominator of $G_{des}(s)$ must be the same as the degree of the denominator of $G(s)$ after cancellation.

(3) The coefficient of the highest degree term in the denominator of $G_{des}(s)$ must be unity.

Problems formulated as in figure 2 are ready to be solved using the computer program. In the discussion of the program, we refer to the following two types of problems:

Case I: The coefficients of the denominator polynomial of $C(s)/R(s)$ are linear in the h_i and no other unknowns appear. Such problems are those for which the time constants of $G(s)$ are all known, the number of state variables being fed back is equal to the number of poles in $G(s)$ after cancellation, and $G_1(s)$ is of the form $k/(s + c)$.

Case II: The coefficients of the denominator polynomial of $C(s)/R(s)$ are nonlinear in h_i , or contain other unknowns. The design procedure may be applied to a $G(s)$ in which one or two of the time constants are not defined; this is a Case II problem. If the number of state variables being fed back is less than the number of poles in $G(s)$ after cancellation, this is also a Case II problem.

DESCRIPTION OF THE PROGRAM

The computer program is written in FORMAC and is executed at Lewis Research Center on a 7094 II-7044 DCS. This program accepts as input (and immediately writes as output) the polynomials in s and possibly unknown time constants A and B which are the numerators and denominators of the G_i 's of figure 2. The program then generates the algebraic expressions for the numerator and denominator of $H_{eq}(s)$ and of $C(s)/R(s)$. The coefficients of these four polynomials are then output. Case II type problems are terminated at this point (see Problem Preparation, Case II, p. 5), since the expressions of h_i in $H_{eq}(s)$ are either nonlinear or contain unspecified quantities (A and B).

For Case I type problems, where the expressions are linear, further input is accepted and printed out. This input must be a set of numbers, the first being a numerical value of k , the closed-loop gain in $G_1(s)$, and the remaining numbers being the numerical coefficients of the denominator of the desired transfer function $G_{des}(s)$. They are used to establish a set of simultaneous linear equations which are then solved for the numerical values of the h_i 's. These feedback coefficients are printed, followed by the zeroes and poles of the closed-loop characteristic equation, by the poles of $C(s)/R(s)$ for several multiples (k') of the input value of k (where $k' = [.1, .5, 1., 1.5, 2., 3., 5.]$), and finally by the zeroes of $C(s)/R(s)$. A root locus diagram terminates the output for such an input set of numbers. If there are further such sets of numbers, the program will repeat the procedure described in this paragraph for each set.

PREPARATION OF INPUT DATA SHEET

Figure 4 is a blank input data sheet to be filled in according to the instructions on input data sheet instruction form (fig. 5). The first line of the input data sheet is available for a title (of not more than 79 characters). The next line must have three two-digit integers; the first (KG) is the number of blocks ($3 \leq KG \leq 10$) in the open-loop transfer function, the second (NN) is the order of the numerator of the open-loop transfer function G after cancellation, the third (ND) is the order of the denominator of the open-loop transfer function after cancellation. Case I problems must have $KG = ND$. For Case II problems ND may be as large as 38.

The following lines have a two-digit sequence number--from 01 through twice KG. The FORMAC expression¹ for the numerator of the first (left-most) block is followed by that for the denominator of the first block, then the numerator of the second block, etc., through the denominator of the last (right-most) block. This is all that is required for a Case II problem.

For Case I problems, further input consists of a single value for k, followed by numbered lines (01 through ND) each of which contains a coefficient of the denominator polynomial of $G_{des}(s)$ in descending order, beginning with the coefficient for the s^{ND-1} term (the coefficient for the s^{ND} term is always unity), and on the ND^{th} line, the constant term.

Further sets of one value of K and ND coefficients may follow.

Figure 6(a) is an input data form completed for the example problem (Case I). Figure 6(b) is an input data form for a sample Case II problem $\left(G(s) = \frac{k(s+B)}{(s+2)(s+4)s} \right)$, created by substituting the dummy variable B into the example problem of figure 1(a) to illustrate the differences in input and output for Cases I and II.

Input data cards are punched directly from the completed input data form. For Case II problems, a single blank card follows the other cards.

DESCRIPTION OF OUTPUT

Figures 7(a) and (b) are the complete output listings from the input supplied in figures 6(a) and (b). The first page displays whatever title has been provided on the first card of the input and is followed by the information from the first group of cards with appropriate identification. Pages 2 and 3 display the coefficients of s in the four polynomials formed by the program -- the numerator and denominator of $H_{eq}(s)$ and the numerator and denominator of $C(s)/R(s)$ -- in terms of s, the input gain k, the h_i 's (feedback coefficients), and the unknown time constants A and B, if used. The output for Case II problems (fig. 7(b)) terminates at this point.

¹A FORMAC expression is similar to a FORTRAN expression in that it uses * for multiplication and ** for exponentiation, etc., but it must be terminated by \$.

The polynomial $s^3 + 7s^2 + 2.2s + 1.4$ is a correct FORMAC expression when written in any one of the following ways:

- (1) $s^{**} 3+7.* s^{**} 2+2.2*s+1.4 \$$
- (2) $s*s+s+7.* s*s+2.2*s+1.4 \$$
- (3) $((s+7.)*s+2.2)*s+1.4 \$$

In figure 7(a), page 4 displays the coefficients of $G_{des}(s)$ -- all but the first of which are input, followed by the input value for CK (the closed-loop gain). The lower half of this page displays the numerical values of the h_i 's that satisfy the equations defined by the input values immediately above. Page 5 displays the values of the zeroes followed by the values of the poles of the characteristic equation. The roots of this equation are then printed for each of several multiples of the input design value of k. The zeroes of $C(s)/R(s)$ follow.

All the points on pages 5 and 6 are plotted in the root locus diagram on the last page of figure 7(a). The legend below the plot shows the number of points of each type, but they may not all appear on the plot because overprinting is not used.

If additional sets of CK and coefficients are provided, the output will repeat from page 4 through the plot for each such set.

Lewis Research Center,
National Aeronautics and Space Administration,
Cleveland, Ohio, May 10, 1967,
125-23-02-10-22.

APPENDIX A

DESIGN OF A FUEL VALVE CONTROL SYSTEM

One of the first applications of the state variable feedback technique was to a trial design of a fuel valve servomechanism for a J-85 jet engine. The engine is being used by Lewis Research Center for studying engine and inlet controls for the supersonic transport.

In order to apply the design technique, it was necessary to use a linear model of the physical system. The system transfer function for the torque motor, hydraulic servo-valve, and fuel valve was found to be

$$G(s) = \left(\frac{2.5 k}{s + 2500} \right) \left(\frac{2.262 \times 10^6 s}{s^3 + 3.669 \times 10^3 s^2 + 2.104 \times 10^7 s + 1.762 \times 10^{10}} \right) \\ \left(\frac{5.769 \times 10^3}{s} \right) \left(\frac{8.483 \times 10^{11}}{s^2 + 6.683 \times 10^3 s + 3.28 \times 10^8} \right) \left(\frac{1}{s} \right) \quad (A1)$$

As is evident, $G(s)$ is a 7th order system after cancellation. This means that seven state variables will completely describe the system and, consequently, in the block diagram $G(s)$ must contain seven blocks. The output of each of these blocks will be a state variable to be fed back. The system $G(s)$ may be described as

$$G(s) = \prod_{i=1}^{i=7} K_i G_i(s)$$

where

$$K_1 G_1(s) = \frac{2.5 k}{s + 2500}$$

$$K_2 G_2(s) = \frac{2.262 \times 10^6 s^2}{s^3 + 3.669 \times 10^3 s^2 + 2.104 \times 10^7 s + 1.762 \times 10^{10}}$$

$$K_3 G_3(s) = 1/s$$

$$K_4 G_4(s) = 5.769 \times 10^3 / s$$

$$K_5 G_5(s) = \frac{8.483 \times 10^{11} s}{s^2 + 6.383 \times 10^3 s + 3.28 \times 10^8}$$

$$K_6 G_6(s) = 1/s$$

$$K_7 G_7(s) = 1/s$$

The transfer functions $K_3 G_3(s)$ and $K_6 G_6(s)$ have been inserted to supply seven state variables. To maintain the same open-loop transfer function, $K_2 G_2(s)$ and $K_5 G_5(s)$ have been multiplied by s .

The block diagram for state variable feedback is shown in figure 8 where

R reference input

k servoamplifier gain

and the state variables are

I torque motor current

\dot{x}_f flapper valve velocity

x_f flapper valve displacement

x_s spool valve displacement

\ddot{c} actuator acceleration

\dot{c} actuator velocity

c actuator position

This diagram provides enough information to fill in lines 1 to 14 on the input data sheet (fig. 9) for this problem.

The next step was the selection of the closed-loop transfer function $G_{des}(s)$ to realize a system response having a bandwidth as large as possible with a small overshoot in response to a step input. Two complex poles of the closed-loop system were chosen to be given by the transfer function

$$G_{1, des}(s) = \frac{3.28 \times 10^8}{s^2 + 6.683 \times 10^3 s + 3.28 \times 10^8} \quad (A2)$$

Equation (A2) corresponds to part of the fixed system, equation (A1); this choice was made because the bandwidth of equation (A2) is far beyond that of the other component transfer functions of the system and so will have little effect on the system bandwidth. The remaining closed-loop pole locations were chosen after a careful study of the normalized step response and frequency response curves of reference 5. This reference gives the normalized transfer function for systems with no zeroes which cause the following performance index to be minimized:

$$I = \int_0^{\infty} t \cdot |e| dt \quad (A3)$$

In this performance index (the integral of time-multiplied absolute value error), e is the difference between the system response to a unit step input and the step itself. This criterion was chosen because it gave the desired transient characteristic (step response) compatible with the bandwidth requirement (frequency response).

The denominator of $G_{des}(s)$ is completed by using the fifth order polynomial in table V of reference 5. The constant numerator of $G_{des}(s)$ is now chosen so that $G_{des}(0) = 1$, assuring zero steady-state error for step inputs. This is done by letting the numerator equal the value of the denominator as $s \rightarrow 0$. Consequently, the closed-loop transfer function becomes

$$G_{des}(s) = \frac{1.025 \times 10^{27}}{(s^2 + 6.683 \times 10^3 s + 3.28 \times 10^8)(s^5 + 2.8 \omega_0^2 s^4 + 5 \omega_0^2 s^3 + 5.5 \omega_0^3 s^2 + 3.4 \omega_0^4 s + \omega_0^5)} \quad (A4)$$

In equation (A4) the value of ω_0 governs the system bandwidth; the value 5000 was chosen to give a system bandwidth of 600 hertz which is well beyond that of the dynamic characteristics of the engine. Substituting this value in equation (A4) gives

$$G_{des}(s) =$$

$$\frac{1.025 \times 10^{27}}{s^7 + 2.06831 \times 10^4 s^6 + 5.4655 \times 10^8 s^5 + 6.1149 \times 10^{12} s^4 + 4.7720 \times 10^{16} s^3 + 2.4283 \times 10^{20} s^2 + 7.1789 \times 10^{23} s + 1.0250 \times 10^{27}}$$

as the desired closed-loop transfer function.

It is also necessary to determine the value of k (the unspecified constant in the first block of fig. 8). Since the numerators of $G(s)$ and $G_{des}(s)$ must be equal, we set

$$2.5 k(2.262 \times 10^6)(5.769 \times 10^3)(8.483 \times 10^{11}) = 1.027 \times 10^{27}$$

giving $CK = 37\ 036$.

We have now all of the information needed to complete the input data sheet (fig. 9). The complete output listing for this problem is given in figure 10.

As a check, the calculated values of the feedback gains h_i (p. 4 of fig. 10) were used in an analog computer simulation of the linearized fuel valve control system. A system bandwidth of 600 hertz and an overshoot less than 10 percent in the step response was achieved.

APPENDIX B

PROGRAM LISTING

```

$IRBMC MAIN
    SYMARG
    ATOMIC H(10),GN(10),GD(10),S,CK,A,B
    LOGICAL CHK
    COMPLEX RGN(10)
    DIMENSION PGN(10),PGD(10),GNN(10),GDD(10),ARRAY(13)
    DIMENSION C(40),D(40),E(40),F(40),G(40)
    DIMENSION DS(40),FS(40),CH(40),CHCK(40)
    DIMENSION X(20, 20), R(10)
    DIMENSION DFTRN(10),HFTRN(10)
    DIMENSTON DLM(130), ARL(130),PP(11),KK(14),PCK(14)
    EQUIVALENCE (KK, PCK )
    DATA PCK(4), PCK(6), PCK(8),PCK(10) /
        1H0, 1HX, 1H+, 1H*   /
    DIMENSION PCTG(10)
    DATA PCTG / .1, .5, 1., 1.5, 2., 3., 5., 0., 0., 0. /
    DATA JTOP / 7 /

    NLAR PARAM
    1( GN(1), GNN(1)),
    2( GN(2), GNN(2)),
    3( GN(3), GNN(3)),
    4( GN(4), GNN(4)),
    5( GN(5), GNN(5)),
    6( GN(6), GNN(6)),
    7( GN(7), GNN(7)),
    8( GN(8), GNN(8)),
    9( GN(9), GNN(9)),
    1(GN(10),GNN(10))

    DLAR PARAM
    1( GD(1), GDD(1)),
    2( GD(2), GDD(2)),
    3( GD(3), GDD(3)),
    4( GD(4), GDD(4)),
    5( GD(5), GDD(5)),
    6( GD(6), GDD(6)),
    7( GD(7), GDD(7)),
    8( GD(8), GDD(8)),
    9( GD(9), GDD(9)),
    1(GD(10),GDD(10))

    PLAR PARAM
    1( H(1), HFTRN(1) ),
    2( H(2), HFTRN(2) ),
    3( H(3), HFTRN(3) ),
    4( H(4), HFTRN(4) ),
    5( H(5), HFTRN(5) ),
    6( H(6), HFTRN(6) ),
    7( H(7), HFTRN(7) ),
    8( H(8), HFTRN(8) ),
    9( H(9), HFTRN(9) ),
    A( H(10), HFTRN(10) )

70 DO 76 I = 1, 40
    LET C(I) = 0.0
    LET D(I) = 0.0
    LET E(I) = 0.0
    LET F(I) = 0.0
    LET G(I) = 0.0
    LET FS(I) = 0.0
76 CONTINUF
80 DO 90 I = 1, 10
    LET GNN(I) = 0.0
    LET GDD(I) = 0.0
    HFTRN(I) = 0.0
90 CONTINUE

```

```

92 READ (5,499)
95 READ(5,498) KG,NN,ND
96 WRITE ( 6, 499 )
      WRITE (6,501) KG,NN,ND
98 IDATA = 0
100 DO 120 J = 1,KG
102 IDATA = IDATA + 1
      IDMIN1 = IDATA - 1
104 READ( 5, 511 ) ID1, ( ARRAY(I), I = 1, 13 )
106 TFI( ID1 - IDATA ) 108,110,108
108 WRITE( 6, 513 ) IDMIN1
      GO TO 1000
110 JSTART = 0
111 LFT GNN(J) = ALGCON ARRAY(1), JSTART
112 WRITE( 6, 530 ) J
      WRITE( 6, 541 ) ( ARRAY(I), I = 1, 13 )
113 IDMIN1 = IDATA
      IDATA = IDATA + 1
114 READ( 5, 511 ) ID1, ( ARRAY(I), I = 1, 13 )
115 TFI( ID1 - IDATA ) 108,116,108
116 JSTART = 0
117 LET GDD(J) = ALGCON ARRAY(1), JSTART
118 WRITE( 6, 551 ) J
      WRITE( 6, 541 ) ( ARRAY(I), I = 1, 13 )
120 CONTINUE
122 LFT PGN(1)=1.
124 DO 126 J = 2,KG
      LET PGN(J)= PGN(J-1)* GN(J)
126 CONTINUE
128 LFT HFQD = PGN(KG)
      LFT CRN = GN(1)*HFQD
      LFT PGD(KG)=1.
130 DO 132 J = 2,KG
      L = KG-J+1
      LFT PGD(L) = PGD(L+1)*GD(L+1)
132 CONTINUE
134 LFT CRDT1= PGD(1)* GD(1)
      LFT HFQN =0.
136 DO 138 J = 1,KG
      LFT HFQN = HFQN +H(J)*PGN(J)*PGD(J)
138 CONTINUE
140 LFT CRDT2 = GN(1)*HEON
      LFT CRD = CRDT1+CRDT2
141 ERASE CRDT2
142 DO 144 J = 1,KG
      ERASE PGD(J),PGN(J)
144 CONTINUE
146 LFT CRNS = SUBST CRN + NLAB,DLAB
      LFT CRNSE = EXPAND CRNS
      ERASE CRN,CRNS
      LFT HEQDS = SUBST HEQD + NLAB,DLAB
      LFT HEQDSF= EXPAND HEQDS
      ERASE HFQD,HEQDS
      LFT CRDS = SURST CRD + NLAB,DLAB
      LFT CRDSF = EXPAND CRDS
      ERASE CRD,CRDS
      LFT HFONS = SURST HFQN + NLAB,DLAB
      LFT HFONSE= EXPAND HEQNS
      ERASE HEON,HEQNS
      LFT CDT1S = SUBST CRDT1,NLAB,DLAB
      LFT CDT1SF = EXPAND CRDT1S
      ERASE CRDT1,CRDT1S
148 N = 0
149 I=1
151 LFT C(I) = COEFF CRDSF,S**N,V1,V2
      I = I+1
      N=V2
154 IF(N.NE.0) GO TO 151
      I=1
156 LFT D(I) = COEFF CRNSE,S**N,V1,V2
      I = I+1
      N=V2
159 IF(N.NE.0) GO TO 156

```

```

      I=1
161 LFT F(I) = COEFF HEQDSE,S**N,V1,V2
      I = I+1
      N=V2
164 IF(N.NE.0) GO TO 161
      I=1
166 LFT F(I) = COEFF HEONSE,S**N,V1,V2
      I = I+1
      N=V2
169 IF(N.NE.0) GO TO 166
      I = 1
171 LFT G(I) = COEFF COT1SF, S**N, V1, V2
      I = I + 1
      N = V2
174 IF( N .NE. 0 ) GO TO 171
      FRASE CRNSF,CRDSE,HEONSE,HEQDSE,COT1SE
176 LET CHK = MATCH ID, C(1), 0.0
      IF( CHK ) GO TO 179
      LFT C( ND+2 ) = C(1)
179 LFT CHK = MATCH ID, D(1), 0.0
      IF( CHK ) GO TO 182
      LET D( NN+2 ) = D(1)
182 LFT CHK = MATCH ID, E(1), 0.0
      IF( CHK ) GO TO 185
      LET F( NN+2 ) = E(1)
185 LFT CHK = MATCH ID, F(1), 0.0
      IF( CHK ) GO TO 188
      LET F( ND+1 ) = F(1)
188 LFT CHK = MATCH ID, G(1), 0.0
      IF( CHK ) GO TO 191
190 LET G( ND + 2 ) = G(1)
191 CALL OUT( KG, NN, ND, C, D, E, F )
1992 DO 1996 I = 1,40
1994 FRASE F(I)
1996 CONTINUE
200 CALL R7 ( KG, CKI, R )
210 DO 250 J=1,KG
      LET TEMP = SUBST C(J+2),(CK,CKI)
      DO 230 K = 1,KG
      LFT TFMP1 = COFFF TEMP,H(K)
      LFT X(J,K) = EVAL TEMP1
      LFT TFMP = SURST TEMP,(H(K),0.0)
230 CONTINUE
      LFT R(J) = EVAL R(J)-TEMP
250 CONTINUE
      FRASE TEMP, TEMP1
260 CALL MATINV( X, KG, KSIG )
262 DO 280 I1 = 1,KG
      HFTRN(I1) = 0.
      DO 280 I2 = 1,KG
      HFTRN(I1) = HFTRN(I1) + X(I1,I2) * R(I2)
280 CONTINUE
282 WRITEF( 6, 590 )
284 DO 288 I = 1, KG
      WRITEF( 6, 592 ) I, HFTRN(I)
288 CONTINUE
290 INDEX = 0
8000 WRITEF( 6, 906 )
8002 NLFSSI = ND - 1
     NPLUS1 = ND + 1
8006 DO 8010 J = 2,NPLUS1
      LFT FS(J) = SURST F(J), PLAB
8010 CONTINUE
8012 LET UNITY = EVAL FS(2)
8014 IF(UNITY.EQ.0.1 GO TO 8050
8020 CALL FMTOFN( NLFSSI, FS, DFTRN )
8022 DO 8026 J = 1,NLESS1
      DFTRN(J) = DFTRN(J) / UNITY
8026 CONTINUE
8028 IF (NLFSSI.EQ.1) GO TO 8082
8030 CALL ROOTXX( DFTRN, RGN, NLESS1 )
8032 GO TO 8090
8050 NLFSSI = NLFSSI-1
8052 IF(NLFSSI.EQ.0) GO TO 8070

```

```

8060 DO 8064 J = 2,11
8062 LFT FS(J-1) = FS(J)
8064 CONTINUE
    GO TO 8012
8070 WRITE(6,560)
    GO TO 8150
8082 RGN(1) = DFTRN(1) *(-1.)
8090 DO 8100 J = 1,NLFSSI
    INDEX = INDEX + 1
    ARL( INDEX ) = REAL( RGN(J) )
    DIM( INDEX ) = AIMAG( RGN(J) )
    WRITE( 6, 902 ) ARL(INDEX), DIM(INDEX)
8100 CONTINUE
8150 WRITE( 6, 904 )
8160 CALL FMTOFN( ND, G, DFTRN )
8170 CALL ROOTXX( DFTRN, RGN, ND )
8180 DO 8200 J = 1, ND
    INDEX = INDEX + 1
    ARL( INDEX ) = REAL( RGN(J) )
    DIM( INDEX ) = AIMAG( RGN(J) )
    WRITE( 6, 902 ) ARL(INDEX), DIM(INDEX)
8200 CONTINUE
8400 WRITE( 6, 908 ) CKI
8402 NPLUS2 = ND + 2
8404 DO 8410 J = 2,NPLUS2
    LET CH(J) = SUBST C(J), PLAB
8410 CONTINUE
8412 DO 8470 J = 1,JTOP
    GAIN = PCTG(J) * CKI
8416 DO 8420 I = 2,NPLUS2
    LET CHCK(I) = SURST CH(I), ( CK, GAIN )
8420 CONTINUE
8430 CALL FMTOFN( ND, CHCK, DFTRN )
8440 CALL ROOTXX( DFTRN, RGN, ND )
8450 DO 8460 K = 1,ND
    INDEX = INDEX + 1
    DIM( INDEX ) = AIMAG( RGN(K) )
    ARL( INDEX ) = REAL( RGN(K) )
    WRITE( 6, 910 ) PCTG(J), GAIN, ARL(INDEX), DIM(INDEX)
8460 CONTINUE
    WRITE( 6, 901 )
8470 CONTINUE
8210 WRITE( 6, 900 )
8220 NPLUS2 = NN + 2
8230 DO 8234 J = 2,NPLUS2
    LFT DS(J) = SUBST D(J), ( CK, 1. )
8234 CONTINUE
8238 IF (NN.FQ.0) GO TO 8270
8246 CALL FMTOFN( NN, DS, DFTRN )
    LFT TFMPD = EVAL DS(2)
    DO 8247 I = 1,NN
8247 DFTRN(I) = DFTRN(I)/TEMPO
8248 IF (NN.FQ.1) GO TO 8282
8250 CALL ROOTXX( DFTRN, RGN, NN )
8252 GO TO 8290
8270 WRITE( 6,560)
8272 GO TO 870
8282 RGN(1) = DFTRN(1) *(-1.)
8290 DO 8300 J = 1,NN
    INDEX = INDEX + 1
    ARL( INDEX ) = REAL( RGN(J) )
    DIM( INDEX ) = AIMAG( RGN(J) )
    WRITE( 6, 902 ) ARL(INDEX), DIM(INDEX)
8300 CONTINUE

```

```

870 PP(1) = 5.
PP(3) = 0.
PP(4) = 0.
KK(1) = 7
KK(2) = 4
KK(3) = NLFSS1
KK(5) = ND
KK(7) = JTDP * ND
KK(9) = NN
WRITEF( 6, 911 )
CALL PLOTMY( DIM, ARL, KK, PP )
WRITEF( 6, 912 ) KK(3),KK(5),KK(7),KK(9)
999 GO TO 200
498 FORMAT(3I2)
499 FORMAT(1H1,79H
4991
501 FORMAT( 1H1, 42X, 31HSTATE VARIABLE FEEDBACK PROGRAM /
5011      1H1, 39X, 36H----- INPUT DATA ----- /
5012      1H1, 19X,
5013      37H THE OPEN LOOP TRANSFER FUNCTION HAS ,I2,8H BLOCKS.   /
5014      1HK, 19X,
5015      45H THE NUMERATOR POLYNOMIAL OF C/R IS OF ORDER ,I2, 1H./
5016      1HK, 19X,
5017      47H THE DENOMINATOR POLYNOMIAL OF C/R IS OF ORDER ,I2,1H./
5018      1HK, 9X, 62HGN(I) AND GD(I) ARE THE NUMERATOR AND DENOMINA
5019TOR OF THF I-TH /
5019  10X,40HBLLOCK COUNTING FROM LEFT TO RIGHT.           )
511 FORMAT( I2, 13A6 )
513 FORMAT(1H1,9X,24HDATA CARD FOLLOWING CARD ,I2,17H IS OUT OF ORDER.
5131)
530 FORMAT( 1H1, 9X, 3HGN(I, I2, 6H) =  )
541 FORMAT( 1H+, 20X, 13A6 )
551 FORMAT( 1HK, 9X, 3HGD(I, I2, 6H) =  )
560 FORMAT(1H0,50X,14HTHERF ARE NONE      )
590 FORMAT( 1H0/
5901 1H0,31X, 54H-----FFEDBACK COEFFICIENTS - H(I) -----
5902/1H0 )
592 FORMAT( 1H0, 9X, 2HH(I, I2, 6H) = , G12.5 )
900 FORMAT( 1H0 / 1H0 / 40X,
9001 39H----- 7FROFS OF C/R ----- / 1H0 /
9002 12X, 9HREAL PART, 10X,14HIMAGINARY PART / 1H0 )
901 FORMAT( 1H )
902 FORMAT( 12X, G12.5, 10X, G12.5 )
904 FORMAT(1H0/1H0/31X, 63H----- POLES OF THE CHARACTERISTIC E
9041 FQUATION ----- /1H0/
9042 12X, 9HREAL PART, 10X,14HIMAGINARY PART / 1H0 )
906 FORMAT(1H1 /1H0 / 32X, 63H----- ZEROS OF THE CHARACTER
9061 STIC FQUATION ----- /1H0 /
9062 12X, 9HREAL PART, 10X, 14HIMAGINARY PART / 1H0 )
908 FORMAT(1H0 /1H0 / 10X, 104H----- ROOTS OF THE CHARACTERI
9081 STIC FQUATION (POLES OF C/R) FOR VARIOUS VALUES OF GAIN -----
9082-- /4X, 28HCK (DESIGN VALUE OF GAIN) = ,G12.5/1H0/7H D/D CK
9083,7X,4HGAIN,13X, 9HREAL PART,10X, 14HIMAGINARY PART / 1H0 )
910 FORMAT( 1H , F6.2, 5X,
9101      G12.5, 7X, G12.5,10X, G12.5 )
911 FORMAT( 2HPT, 57X, 41H----- ROOT LOCUS PLOT ----- )
912 FORMAT(2HPL,48X,2H (.I1,3H) . 48H0 DENOTES A ZERO OF THE CHARAC
9121 TERISTIC EQUATION /
9122      2HPL,48X,2H (.I1,3H) . 48HX DENOTES A POLE OF THE CHARAC
9123 TFRISTIC EQUATION /
9124      2HPL,48X,1H( .I2,3H) . 48H+ DENOTES A ROOT OF THE CHARAC
9125 TFRISTIC EQUATION /
9125      2HPL,48X,2H (.I1,3H) . 24H* DENOTES A ZERO OF C/R )
1000 STOP
END

```

```

$TRMNC OUTNM
      SUBROUTINE OUT( N, NN, ND, C, D, E, F )
C      THIS SUBROUTINE WRITES C/R AND HEQ
      DIMENSION C(40), D(40), E(40), F(40)
      SYMARG      C,D,E,F
      DIMENSION KOUT(21)
      DIMENSION HFTRN(20)
      WRITE( 6, 501 )
      GO TO 752
700  WRITE( 6, 510 )
      NPLUS2 = NN + 2
710  DO 720 L = 2, NPLUS2
      LPOWER = NPLUS2- L
      WRITE( 6, 521 ) LPOWER
      LET DEL = ORDER D(L), INC, FUL
      SIG = 0.
722  LET SIG = BCDCON DEL , KOUT, 21
      WRITE( 6, 531 ) ( KOUT(J), J = 2, 21 )
718  IF( SIG .NE. 0. ) GO TO 722
720  CONTINUE
      WRITE( 6, 540 )
      NPLUS2 = ND + 2
    DO 750 L = 2, NPLUS2
      LPOWER = NPLUS2- L
      WRITE( 6, 521 ) LPOWER
      LET DEL = ORDER C(L), INC, FUL
      SIG = 0.
741  LET SIG = BCDCON DEL , KOUT, 21
      WRITE( 6, 531 ) ( KOUT(J), J = 2, 21 )
748  IF( SIG .NE. 0. ) GO TO 741
750  CONTINUE
      RETURN
752  WRITE( 6, 560 )
      NPLUS2 = ND + 1
    DO 770 L = 2, NPLUS2
      LPOWER = NPLUS2- L
      WRITE( 6, 521 ) LPOWER
      LET DEL = ORDER F(L), INC, FUL
      SIG = 0.
761  LET SIG = BCDCON DEL , KOUT, 21
      WRITE( 6, 531 ) ( KOUT(J), J = 2, 21 )
763  IF( SIG .NE. 0. ) GO TO 761
770  CONTINUE
      WRITE( 6, 580 )
      NPLUS2 = NN + 2
    DO 790 L = 2, NPLUS2
      LPOWER = NPLUS2- L
      WRITE( 6, 521 ) LPOWER
      LET DEL = ORDER E(L), INC, FUL
      SIG = 0.
781  LET SIG = BCDCON DEL , KOUT, 21
      WRITE( 6, 531 ) ( KOUT(J), J = 2, 21 )
783  IF( SIG .NE. 0. ) GO TO 781
790  CONTINUE
      GO TO 700
501  FORMAT( 1H1 / 1H0 /
5011          29X, 59H-----HEQ AND C/R IN TERMS OF THE H(I)
5012          ----- )
510  FORMAT( 1H1,47X,30H(NUMERATOR POLYNOMIAL OF C/R) /1H0)
521  FORMAT( 1HJ, 8X,19H COEFFICIENT OF S**, [2, 5H = )
531  FORMAT( 36X, 16A6 )
540  FORMAT( 1HK /
5401     1HK,19X, 60HCHARACTERISTIC POLYNOMIAL - (DENOMINATOR POLYNOMIAL
5402OF C/R) /1H0)
560  FORMAT(1HL /1H0/
5601     20X, 29HNUMFRATOR POLYNOMIAL OF HEQ /1H0 )
580  FORMAT(1HO /
5801     1HK,19X, 31HDENOMINATOR POLYNOMIAL OF HEQ /1H0) /
      END

```

```

$1BFTC R7NAM
      SUBROUTINE R7( N, CKI, R )
C   THIS SUBROUTINE READS AND WRITES R(I) AND READS AND WRITES CK,
C   USING N = KG.
      DIMENSION R(20)
      READ( 5,530 ) CKI
      TDATA = 0
      DO 10 I = 1, N
      TDATA = TDATA + 1
      IDMIN1 = TDATA - 1
      READ( 5,520 ) ID1, R(I)
      IF( ID1 - TDATA ) 21, 10, 21
21  WRITE( 6,522 ) IDMIN1
      GO TO 62
10  CONTINUE
      WRITE( 6,540 ) N
      DO 50 J = 1, N
      JJ = N - J
      WRITE( 6,543 ) JJ, R(J)
50  CONTINUE
      WRITE( 6,551 ) CKI
50  CONTINUE
      RETURN
520 FORMAT( 12, F13.6 )
522 FORMAT( 1H1, 9X, 24H DATA CARD FOLLOWING CARD ,12, 17H IS OUT OF ORDER.
5221      )
530 FORMAT( F15.6 )
540 FORMAT( 1H1 /
54011H , 31X, 59H----- DESIRED CHARACTERISTIC POLYNOMIAL -----
5402--- / 58X, 7H( INPUT)
5403 / 1H0 /
5404      1HK, 8X, 19H COEFFICIENT OF S**, 12, 9H =    1.0      )
543 FORMAT( 1HK, 8X, 19H COEFFICIENT OF S**, 12, 5H =    , G12.5      )
551 FORMAT( 1HL, 9X, 12HGAIN CK =    , G12.5      )
62 STOP
END

```

```

$TRFTC ROOTR
      SUBROUTINE ROOTXX(A,R,NN)
C SUBROUTINE TO SOLVE POLYNOMIALS WITH NO ROOTS OF ORDER .GT.=2
      COMPLEX CMPLX,CONJG,C SORT,CZERO,CONF,A11,A12,A21,
      1A22,A31,A32,S1,S2,X,DX,TC,F0,F1,F2,R
      COMPLEX X2,TEMP
      DIMENSION A(1),R(1)
      FOUTVALFNCF (F0,A12),(F1,A22),(F2,A31),(TC,A11)
      1,(DX,A11),(S2,A32)
      DATA CZERO,CONF/(0.0,0.0),(1.0,0.0)/
      1,MAX,FPS/35,1.0E-7/
      N=NN
      DO 30 K=1,NN
      TF(A(N),NF,0.0) GO TO 40
      N=N-1
      30 R(K)=CZERO
      RETURN
      40 TF(N,F0,1) GO TO 150
      TF(N,F0,2) GO TO 160
      FN=F1.DAT(N)
      NP1=N+1
      NM1=N-1
      FNM1=F1.DAT(NM1)
      NP1=N+1
      KS=K
      ST1=A(N-1)/A(N)
      ST2=ST1*ST1-(A(N-2)+A(N-2))/A(N)
      ASSIGN 110 TO M1
      45 TF(K,GT,NN) RETURN
      KM1=K-1
      ASSIGN 70 TO M2
      DO 120 L=1,MAX
      GO TO M2,(70,50)
      50 A12=CONF
      A22=CONF
      A32=CONF
      DO 60 T=1,NM1
      A11=A12
      A12=A(T)+X*A11
      A21=A22
      A22=A12+X*A21
      A31=A32
      60 A32=A22+X*A31
      F2=A31+A31
      F0=A(N)+X*A12
      IF(RFAL(F0),F0,0.0,AND,AIMAG(F0),F0,0.0) GO TO 130
      S1=F1/F0
      S2=S1*S1-F2/F0
      GO TO 80
      70 X=CZERO
      S1=CMPLX(ST1,0.0)
      S2=CMPLX(ST2,0.0)
      ASSIGN 50 TO M2
      80 GO TO M1,(110,90)
      90 DO 100 T=KS,KM1
      TC=1.0/(X-R(T))
      S1=S1-TC
      100 S2=S2-TC*TC
      110 IF(RFAL(S1),F0,0.0,AND,AIMAG(S1),F0,0.0) GO TO 115
      TC=FN/S1
      DX=TC/(1.0+CSORT(FNM1*(TC*S2/S1-1.0)))
      GO TO 116

```

```

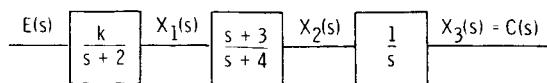
115 DX=FN/CSORT(FNM1*(FN*S2))
116 X=X-DX
   IF(REAL(CABS(DX))/REAL(CABS(X)).LE.EPS)GO TO 130
120 CONTINUE
   NN=KM1
   CALL ARERR(44H SUBROUTINE ROOTXX HAS FAILED TO CONVERGE. $)
130 ASSIGN 90 TO M1
   R(K)=X
   K=K+1
   IF(AIMAG(X).EQ.0.0.DR.K.GT.NN)GO TO 45
   R(K)=CONJG(X)
   K=K+1
   GO TO 45
150 R(K)=CMPLX(-A(1),0.)
   RETURN
160 X1=-.5*A(1)
   TEMP=A(1)*A(1)-4.*A(2)
   X2=.5*CSORT(TEMP)
   R(K-1)=X1+X2
   R(K+1)=X1-X2
   RETURN
   END

$TRFMC FC2FN
SUBROUTINE FMTOFN( NA, A, AFTRN )
DIMENSTON A(10)
SYMARG A
DIMENSTON AFTRN(10)
DO 100 J = 1, NA
   LFT AFTRN(J) = EVAL A( J + 2 )
100 CONTINUE
RETURN
END

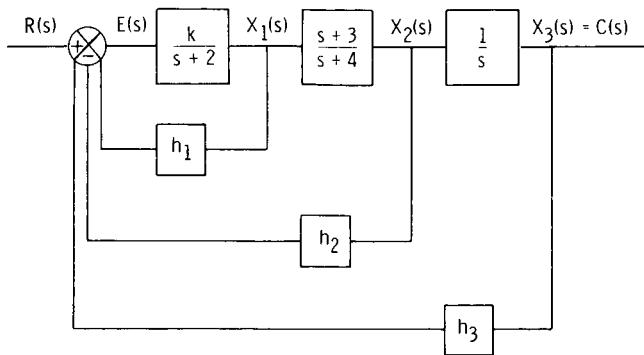
```

REFERENCES

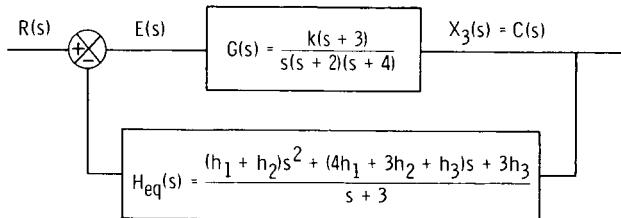
1. Schultz, D. G.; and Melsa, J. L.: State Functions and Linear Control Systems. McGraw-Hill Book Co., Inc., 1967.
2. Kalman, R. E.: When is a Linear Control System Optimal? J. Basic Eng., vol. 86, no. 1, Mar. 1964, pp. 51-60.
3. Brockett, Roger W.: Poles, Zeroes, and Feedback: State Space Interpretation. IEEE Trans. on Automatic Control, vol. AC-10, no. 2, Apr. 1965, pp. 129-135.
4. Schultz, Donald G.: Control System Design by State Variable Feedback Techniques. Vol. 1, Arizona University (NASA CR-77901), July 1966.
5. Graham, D.; and Lathrop, R. C.: The Synthesis of "Optimum" Transient Response: Criteria and Standard Forms. AIEE Trans., Part II, Applications and Industry, vol. 72, no. 9, Nov. 1953, pp. 273-288.



(a) Open-loop system.



(b) Closed-loop system with state variable feedback.



(c) Closed-loop system with equivalent feedback transfer function.

Figure 1. - Example problem.

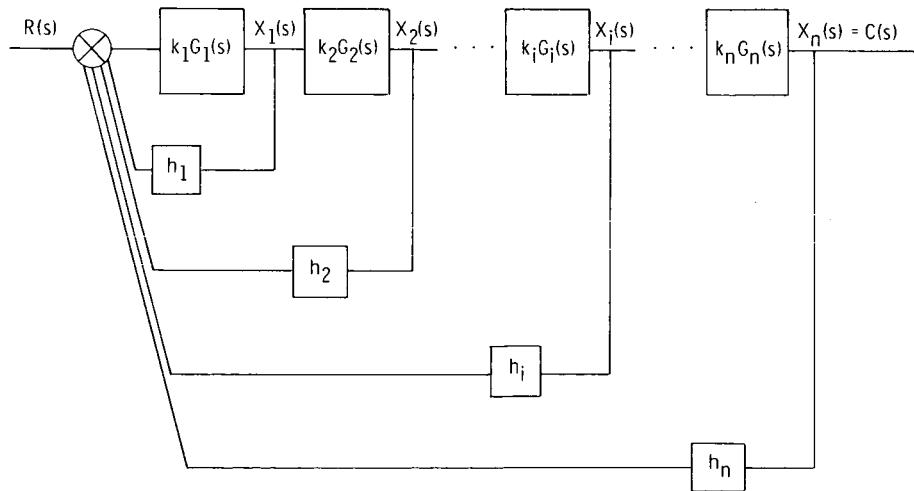
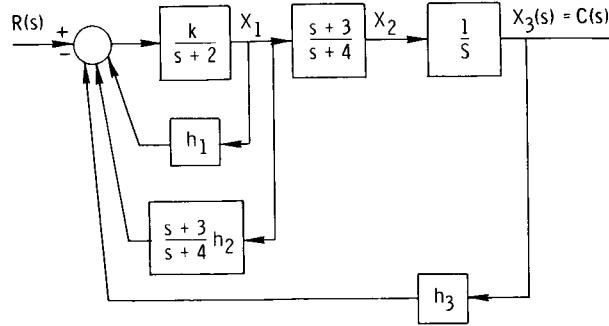
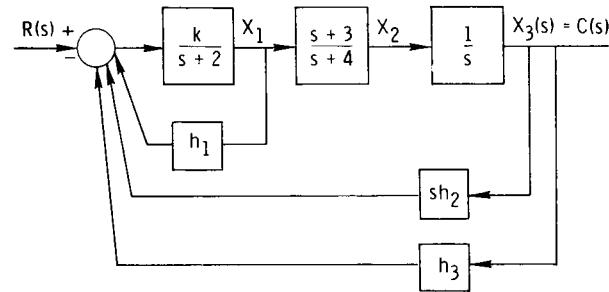


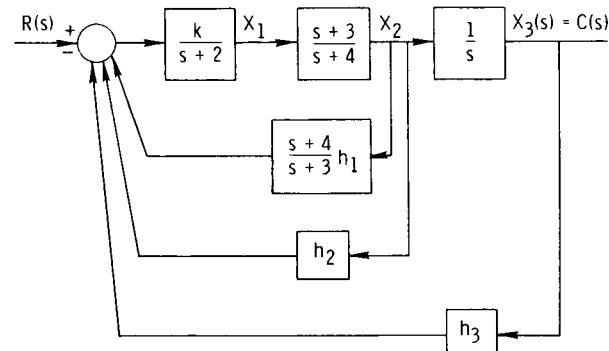
Figure 2. - Generalized block diagram for use in supplying program input data.



(a) Generation of $X_2(s)$ from $X_1(s)$.



(b) Generation of $X_2(s)$ from $X_3(s)$.



(c) Generation of $X_1(s)$ from $X_2(s)$.

Figure 3. - Generation of state variables by block diagram manipulation.

Input Data Sheet - State Variable Feedback

1	2	3	4	5	6	7	8	9	10	11	12	13	14	15			16 to 80
<u>Title (up to 79 characters)</u>																	
1																	
KG	NN	ND															
0	1																
0	2																
0	3																
0	4																
0	5																
0	6																
0	7																
0	8																
0	9																
1	0																
1	1																
1	2																
1	3																
1	4																
1	5																
1	6																
1	7																
1	8																
1	9																
2	0																
<u>±</u>																	
0	1																
0	2																
0	3																
0	4																
0	5																
0	6																
0	7																
0	8																
0	9																
1	0																

Case I problems only

Figure 4. - Input data sheet.

Input Data Sheet Instructions

Figure 5. - Instructions for filling out input data sheet.

1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	← 16 to 80 →	
Title (up to 79 characters)																
1 Example Problem Case I																
KG	NN	ND														
0 3	0 1	0 3														
0 1	CKB															
0 2	x + 2.8															
0 3	x + 3.8															
0 4	x + 4.8															
0 5	1.8															
0 6	x.8															
0 7																
0 8																
0 9																
1 0																
1 1																
1 2																
1 3																
1 4																
1 5																
1 6																
1 7																
1 8																
1 9																
2 0																
±	14.1667															
0 1	8															
0 2	23.5															
0 3	42.5															
0 4																
0 5																
0 6																
0 7																
0 8																
0 9																
1 0																

} Case I problems only

(a) Case I.

Figure 6. - Example problem input data sheet.

Input Data Sheet - State Variable Feedback

(b) Case II.

Figure 6. - Concluded.

EXAMPLE PROBLEM CASE 1

STATE VARIABLE FEEDBACK PROGRAM

----- INPUT DATA -----

THE OPEN LOOP TRANSFER FUNCTION HAS 3 BLOCKS.
THE NUMERATOR POLYNOMIAL OF C/R IS OF ORDER 1.
THE DENOMINATOR POLYNOMIAL OF C/R IS OF ORDER 3.
G_N(1) AND G_D(1) ARE THE NUMERATOR AND DENOMINATOR OF THE 1-TH
BLOCK COUNTING FROM LEFT TO RIGHT.

```
GN( 1 ) = CK $  
GD( 1 ) = S + 7. $  
  
GN( 2 ) = S + 3. $  
GD( 2 ) = S + 4. $  
  
GN( 3 ) = 1. $  
GD( 3 ) = S $
```

(a) Case I.

Figure 7. -- Example problem output listing.

-----HFO AND C/R IN TERMS OF THE H(1) -----

NUMERATOR POLYNOMIAL OF HFO

```
COEFFICIENT OF S** 2 = H(1)+H(2)$  
COEFFICIENT OF S** 1 = 4.0*H(1)+3.0*H(2)+H(3)$  
COEFFICIENT OF S** 0 = 3.0*H(3)$
```

DENOMINATOR POLYNOMIAL OF HFO

```
COEFFICIENT OF S** 1 = 1.0$  
COEFFICIENT OF S** 0 = 3.0$
```

(a) Continued.

Figure 7. - Continued.

3
{NUMERATOR POLYNOMIAL OF C/R}

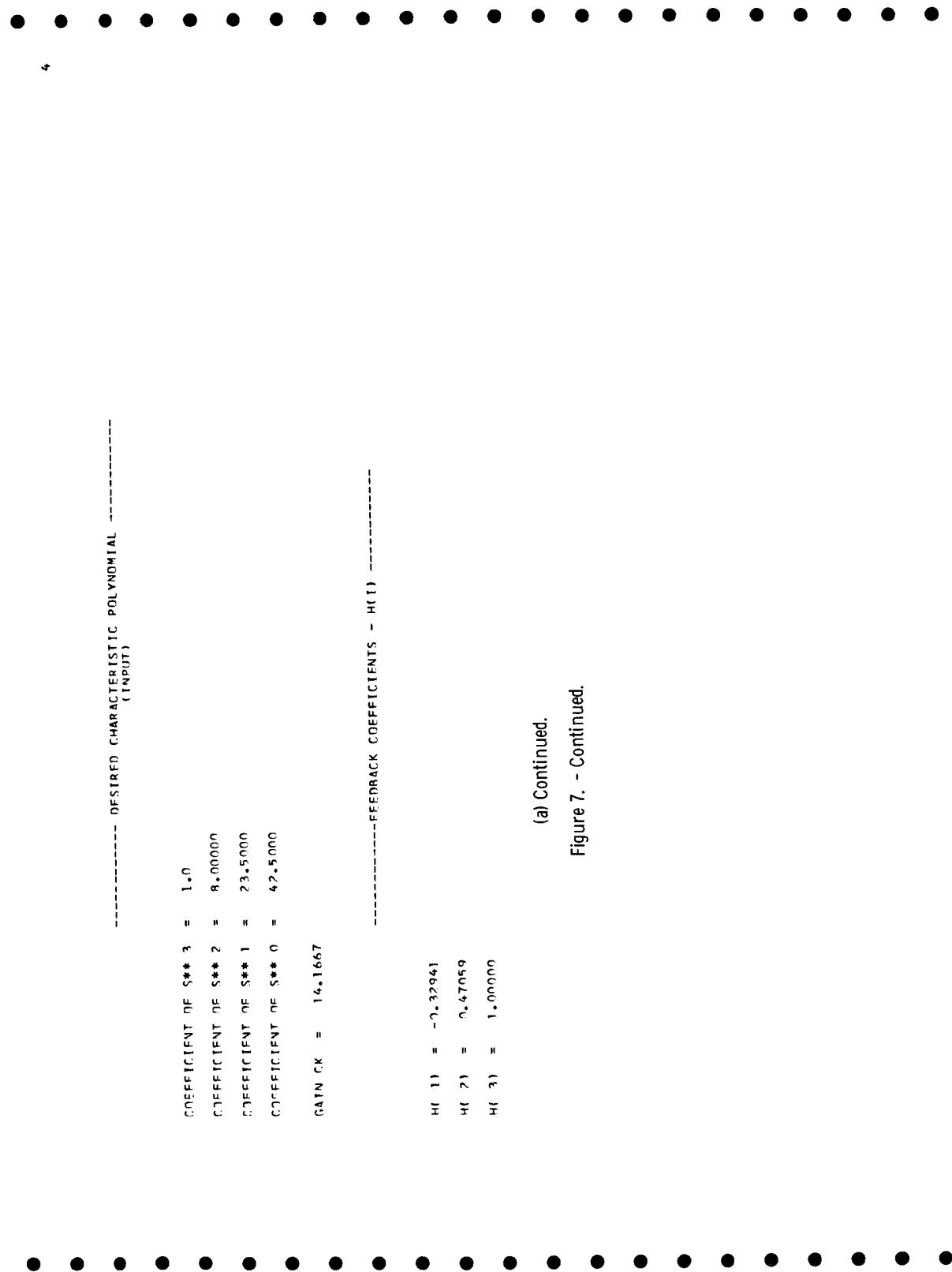
```
C0EFFICIENT OF S** 1 = CK$  
C0EFFICIENT OF S** 0 = 3.0*CK$
```

CHARACTERISTIC POLYNOMIAL - {DENOMINATOR POLYNOMIAL OF C/R}

```
C0EFFICIENT OF S** 3 = 1.0$  
C0EFFICIENT OF S** 2 = CK*H(1)+CK*H(2)+6.0$  
C0EFFICIENT OF S** 1 = 4.0*CK*H(1)+3.0*CK*H(2)+CK*H(3)+8.0$  
C0EFFICIENT OF S** 0 = 3.0*CK*H(3)$
```

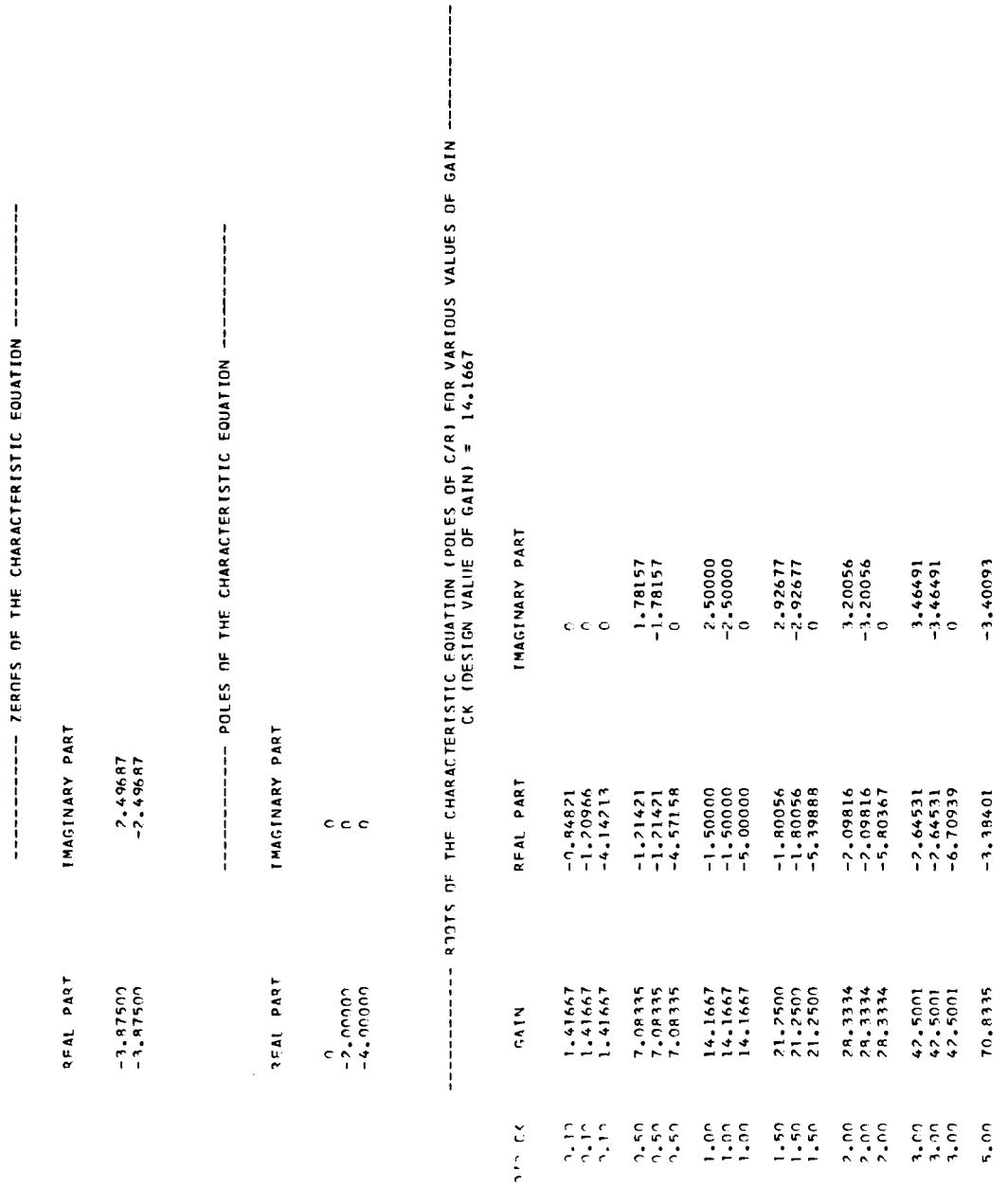
(a) Continued.

Figure 7. - Continued.



(a) Continued.

Figure 7. - Continued.



(a) Continued.

Figure 7. - Continued

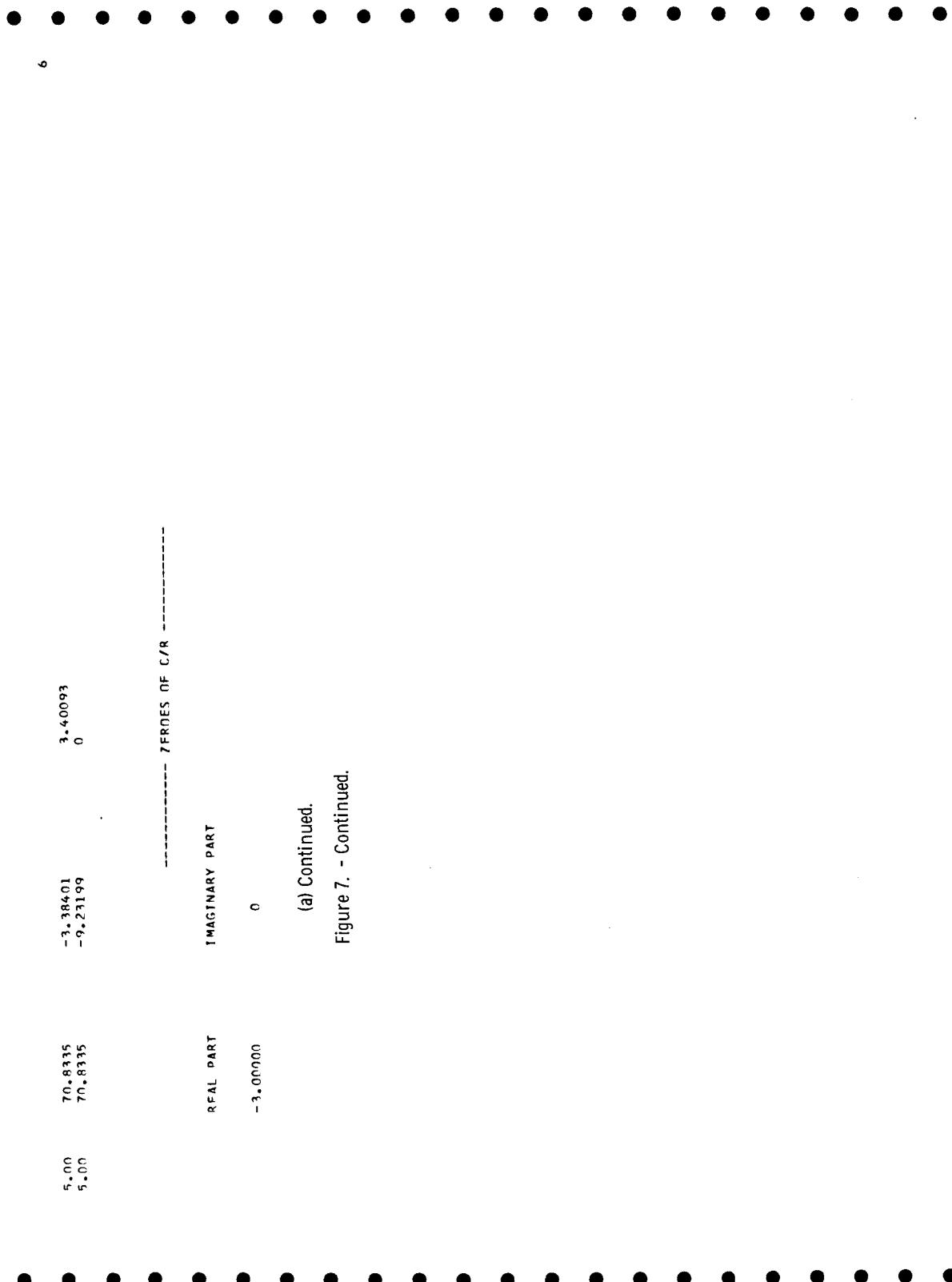
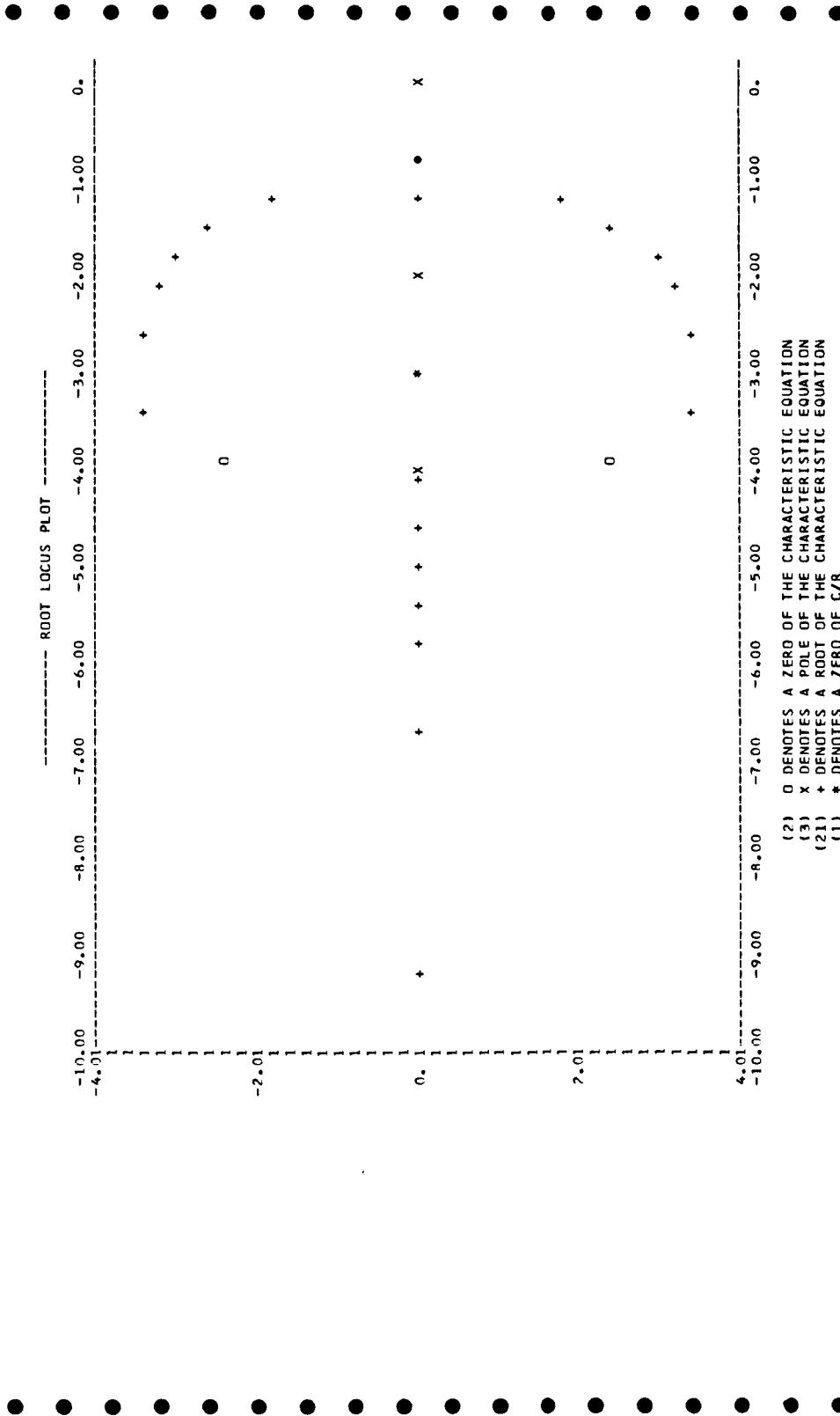


Figure 7. - Continued.



(a) Concluded.

Figure 7. – Continued.

EXAMPLE PRNLFM CASE II

STATE VARIABLE FEEDBACK PROGRAM

----- INPUT DATA -----

THE OPEN LOOP TRANSFER FUNCTION HAS 3 BLOCKS.

THE NUMERATOR POLYNOMIAL OF C/R IS OF ORDER 1.

THE DENOMINATOR POLYNOMIAL OF C/R IS OF ORDER 3.

GN(1) AND GD(1) ARE THE NUMERATOR AND DENOMINATOR OF THE 1-TH
BLOCK COUNTING FROM LEFT TO RIGHT.

$$GN(1) = CK \$$$

$$GD(1) = S + 2. \$$$

$$GN(2) = S+8 \$$$

$$GD(2) = S + 4. \$$$

$$GN(3) = 1. \$$$

$$GD(3) = S \$$$

(b) Case II.

Figure 7. - Continued.

2

THE AND C/R IN TERMS OF THE HII

NUMERATOR POLYNOMIAL OF HEG

```

CNEFFICIENT OF S** 2 = H(1)+H(2)$
CNEFFICIENT OF S** 1 = B+H(2)+4.0*H(1)+H(3)$
CNEFFICIENT OF S** 0 = R*H(3)$

```

DENOMINATIONS AND SYNONYMS OF HEG

COEFFICIENT OF $S^{**} 1 = 1.05$
COEFFICIENT OF $S^{**} 0 = 26$

(b) Continued

Figure 7 Continued

(NUMERATOR POLYNOMIAL OF C/R)

```
C0EFFICIENT OF S** 1 = CK$  
C0EFFICIENT OF S** 0 = B*CK$
```

CHARACTERISTIC POLYNOMIAL - (DENOMINATOR POLYNOMIAL OF C/R)

```
Coefficient of S** 3 = 1.0$  
Coefficient of S** 2 = CK*H(1)*CK*H(2)+6.0$  
Coefficient of S** 1 = R*CK*H(2)+4.0*CK*H(1)*CK*H(3)+8.0$  
Coefficient of S** 0 = B*CK*H(3)$
```

(b) Concluded.

Figure 7. - Concluded.

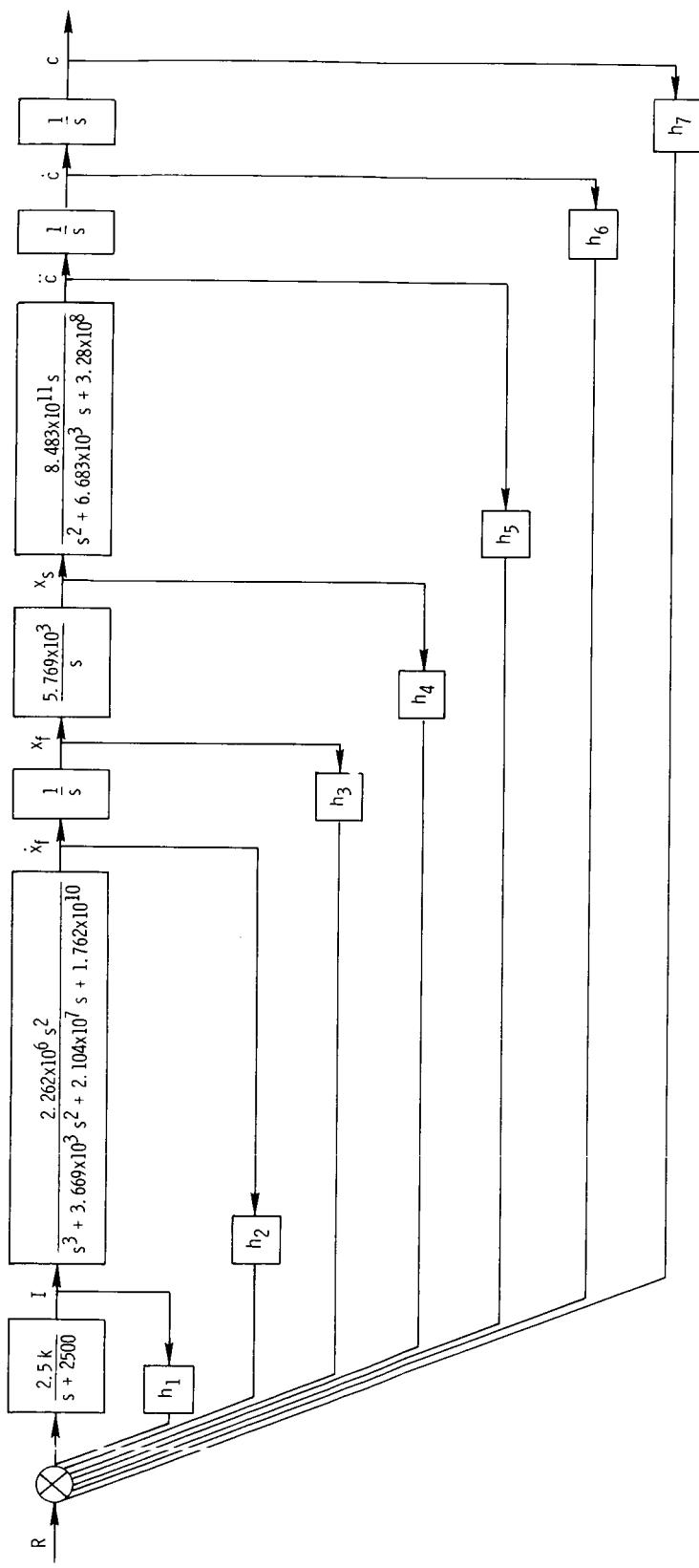


Figure 8. - Fuel valve control system showing state variable feedback.

Input Data Sheet - State Variable Feedback

1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	←	→	16 to 80		
Title (up to 79 characters)																			
1 Appendix A Problem																			
KG	NN	ND																	
07	00	07																	
0 1	$C \neq + 2.58$																		
0 2	$S + 2500. \$$																		
0 3	$2.262E6 * S^{**2} \$$																		
0 4	$S^{**3} + 3.669E3 * S^{**2} + 2.104E7 * S + 1.762E10 \$$																		
0 5	1. \\$																		
0 6	S \\$																		
0 7	5.769E3 \\$																		
0 8	S \\$																		
0 9	8.483E11 * S \\$																		
1 0	$S^{**2} + 6.683E3 * S + 3.28E8 \$$																		
1 1	1. \\$																		
1 2	S \\$																		
1 3	1. \\$																		
1 4	S \\$																		
1 5																			
1 6																			
1 7																			
1 8																			
1 9																			
2 0																			
+ 37036.																			
0 1	20683.1																		
0 2	.54656E9																		
0 3	.61149E13																		
0 4	.47720E17																		
0 5	.24283E21																		
0 6	.71789E24																		
0 7	.10250E28																		
0 8																			
0 9																			
1 0																			

Case I problems only

Figure 9. - Appendix A problem input data sheet.

APPENDIX A PROFLFM

STATE VARIABLE FEEDBACK PROGRAM

----- INPUT DATA -----

THE RIFPN LOOP TRANSFER FUNCTION HAS 7 BLOCKS.
 THE NUMERATOR POLYNOMIAL OF C/R IS OF ORDER 0.
 THE DENOMINATOR POLYNOMIAL OF C/R IS OF ORDER 7.
 GNI(1) AND GDI(1) ARE THE NUMERATOR AND DENOMINATOR OF THE 1-TH
 BLOCK COUNTING FROM LEFT TO RIGHT.

```

GNI( 1 ) = 2.5 * CK $
GDI( 1 ) = S + 2500. $

GNI( 2 ) = 2.26256 * S**2 $
GDI( 2 ) = S**3 + 3.669 F3 * S**2 + 2.104 E7 * S + 1.762 E10 $

GNI( 3 ) = 1.0 $
GDI( 3 ) = S $

GNI( 4 ) = 5.769 F3 $
GDI( 4 ) = S $

GNI( 5 ) = R.493F11 * S $
GDI( 5 ) = S**7 + 6.683 E3 * S + 3.28E8 $

GNI( 6 ) = 1.0F0 $
GDI( 6 ) = S $

GNI( 7 ) = 1.0F0 $
GDI( 7 ) = S $

```

Figure 10. - Appendix A problem output listing.

```

-----HFO AND C/R IN TERMS OF THE H(1) -----
-----  

NUMERATOR POLYNOMIAL OF HFO
  

COEFFICIENT OF $** 6 = H(1)$
  

COEFFICIENT OF $** 5 = 1.0352*0*H(1)+2*62000.0*H(2)$
  

COEFFICIENT OF $** 4 = 3.73559922E8H(1)*l.5116946E10*H(2)+2262000.0*H(3)$
  

COEFFICIENT OF $** 3 = 1.36166232E12*H(1)+7.41935998E14*H(2)+1.5116946E10*H(3)+1.30494779E10*H(4)$
  

COEFFICIENT OF $** 2 = 7.0188744E15*H(1)+7.41935998E14*H(3)+8.72096613E13*H(4)+1.10698721E22*H(5)$
  

COEFFICIENT OF $** 1 = 5.77935994E18*H(1)+4.28022876E19*H(4)+1.10698721E22*H(6)$
  

COEFFICIENT OF $** 0 = 1.10698721E22*H(7)$
  

DENOMINATOR POLYNOMIAL OF HFO
  

COEFFICIENT OF $** 0 = 1.10698721E22$
```

Figure 10. - Continued.

{ NUMERATOR POLYNOMIAL OF C/R }

$$\text{COEFFICIENT OF } S^{**} 0 = 2.76746802E27*CK\$$$

CHARACTERISTIC POLYNOMIAL - {DENOMINATOR POLYNOMIAL OF C/R}

COEFFICIENT OF $S^{**} 7$	=	1.0\$
COEFFICIENT OF $S^{**} 6$	=	2.4999998*CK*H(1)+12852.0\$
COEFFICIENT OF $S^{**} 5$	=	25880.0*CK*(H(1)+5655000.0*CK*H(2))+3.99439923E8\$
COEFFICIENT OF $S^{**} 4$	=	9.3369992E8*CK*H(1)+3.77923669E10*CK*H(2)+5655000.0*CK*H(3)+2.2955621E12\$
COEFFICIENT OF $S^{**} 3$	=	3.40415575E12*CK*H(1)+1.85483998E15*CK*H(2)+3.77923649E10*CK*H(3)+3.26236348E10*CK*H(4)+1.0423030E16\$
COEFFICIENT OF $S^{**} 2$	=	1.75471861E16*CK*H(1)+1.85483998E15*CK*H(3)+2.18124153E14*CK*H(4)+2.76746802E22*CK*H(5)+2.33265458E19\$
COEFFICIENT OF $S^{**} 1$	=	1.44484E19*CK*H(1)+1.07005719E19*CK*H(4)+2.76746802E22*CK*H(6)+1.44483997E22\$
COEFFICIENT OF $S^{**} 0$	=	2.76746802E22*CK*H71\$

Figure 10. - Continued.

```

----- DESIRED CHARACTERISTIC POLYNOMIAL -----  

{INPUT}

COEFFICIENT OF S** 7 = 1.0  

COEFFICIENT OF S** 6 = 20683.1  

COEFFICIENT OF S** 5 = 0.5665E 09  

COEFFICIENT OF S** 4 = 0.61149E 13  

COEFFICIENT OF S** 3 = 0.47720E 17  

COEFFICIENT OF S** 2 = 0.24283E 21  

COEFFICIENT OF S** 1 = 0.71789E 24  

COEFFICIENT OF S** 0 = 0.10250E 28  

GAIN CK = 37036.0  

----- FEEDBACK COEFFICIENTS - H(1) -----  

H( 1) = 0.84578E-01  

H( 2) = 0.31519E-03

```

Figure 10. - Continued.

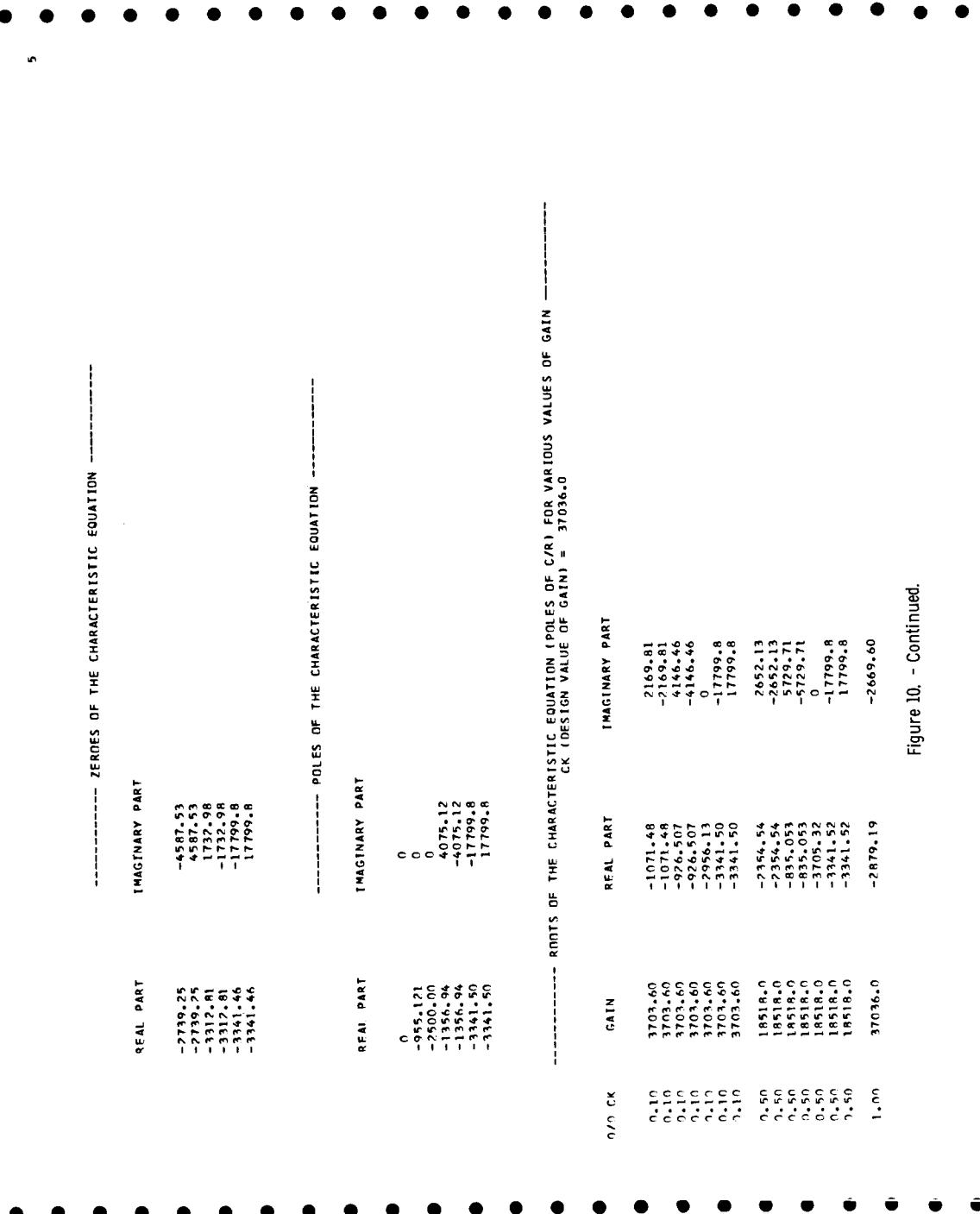


Figure 10. - Continued.

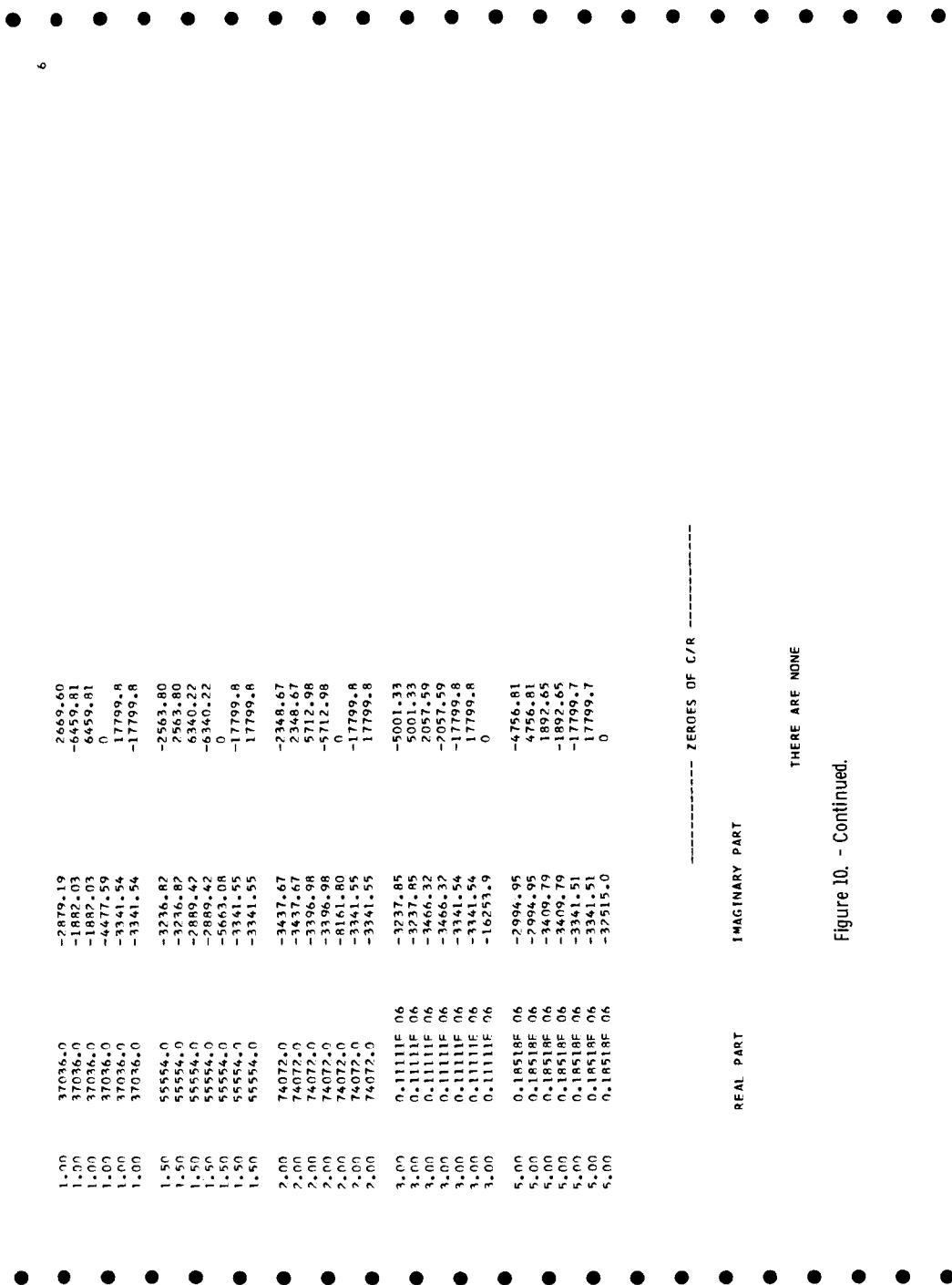


Figure 10. - Continued.

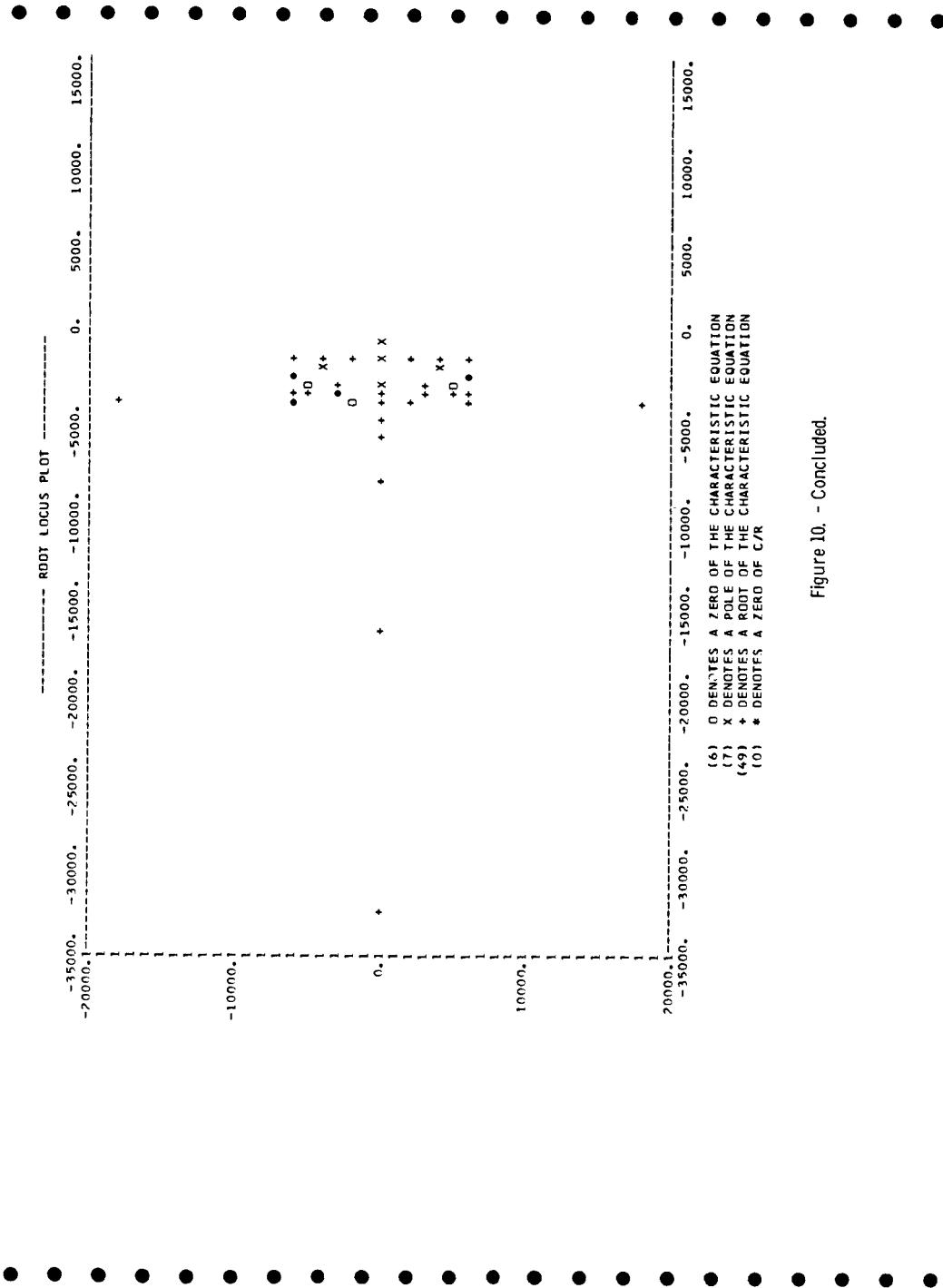


Figure 10. - Concluded.